

# Workshop on Wind Energy Computational Analyses

## Wind Turbine Rotor CFD Analyses by using SU2

Ali Ata ADAM, Nilay SEZER-UZOL  
Dept. of Aerospace Engineering, METU

21 September 2022

# Wind Energy Computational Analyses

EuroCC@Türkiye & METU RUZGEM Workshop



## Content:

1. Wind Turbine Rotor Aerodynamics and CFD Implementations
2. SU2 Capabilities and Tutorial
3. Applications

## Content

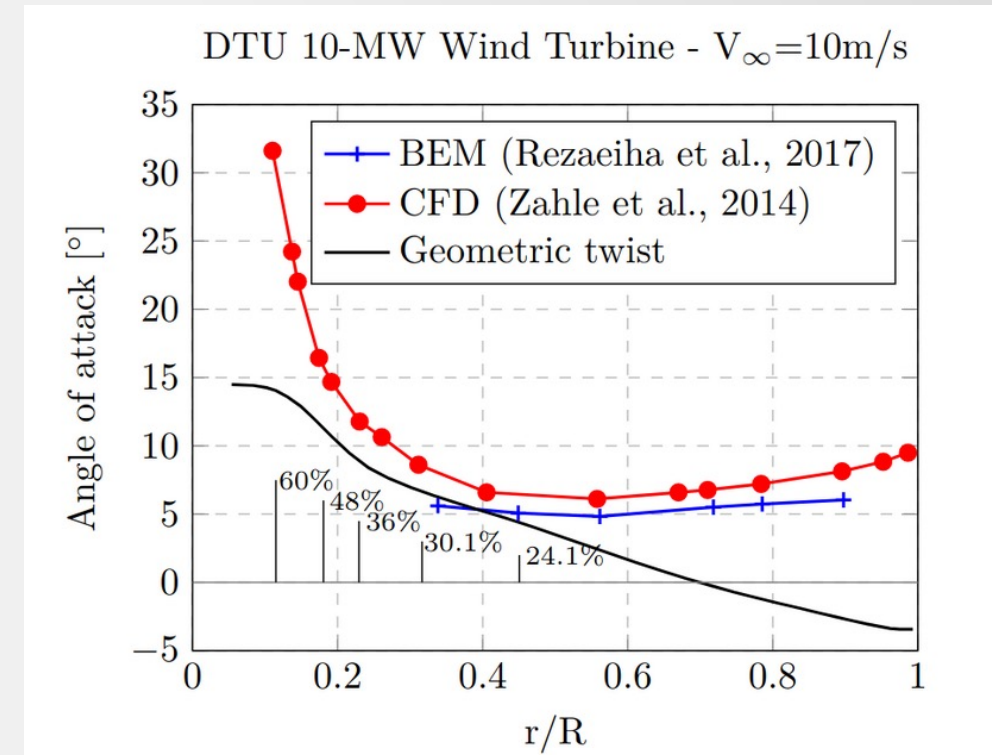
1. Wind Turbine Rotor Aerodynamics and CFD Implementations
  1. Flow Over the Blade
  2. Interactions with Turbine Components
  3. Interactions with Non-Turbine Elements
  4. Wakes
  5. Additional Considerations
2. SU2 Capabilities and Tutorial
  1. Fundamentals
  2. Dynamic Mesh
  3. Flow Over the Blade
  4. Interactions with Rotor Components
  5. Actuator Disk
  6. Surface Roughness and Gusts
3. Applications
  1. Test Case Configuration
  2. SU2 Configuration
  3. Results
  4. Possible Improvements

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Flow Over the Blade

For a commercial wind turbine,

- $M_{tip} \approx 0.2$ : Incompressibility persists but compressibility is knocking on the door.
- Angle of attack: Exceeding the stall angle near the root, in the linear-regime at the mid and tip sections
- Reynolds number
- Separation
- Transition



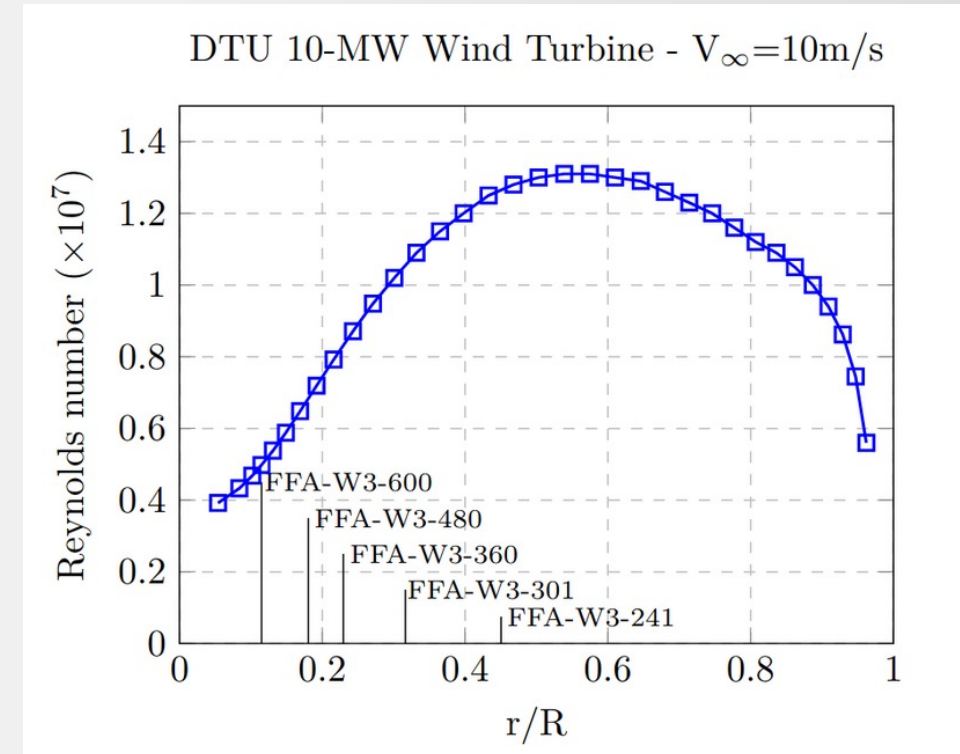
*Variation of angle of attack on the DTU 10MW wind turbine blade operating at a wind speed of 10 m/s*

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Flow Over the Blade

For a commercial wind turbine,

- Compressibility
- Angle of attack
- Reynolds number: In the order of 10 million => high (turbulence) resolution requirement
- Separation
- Transition



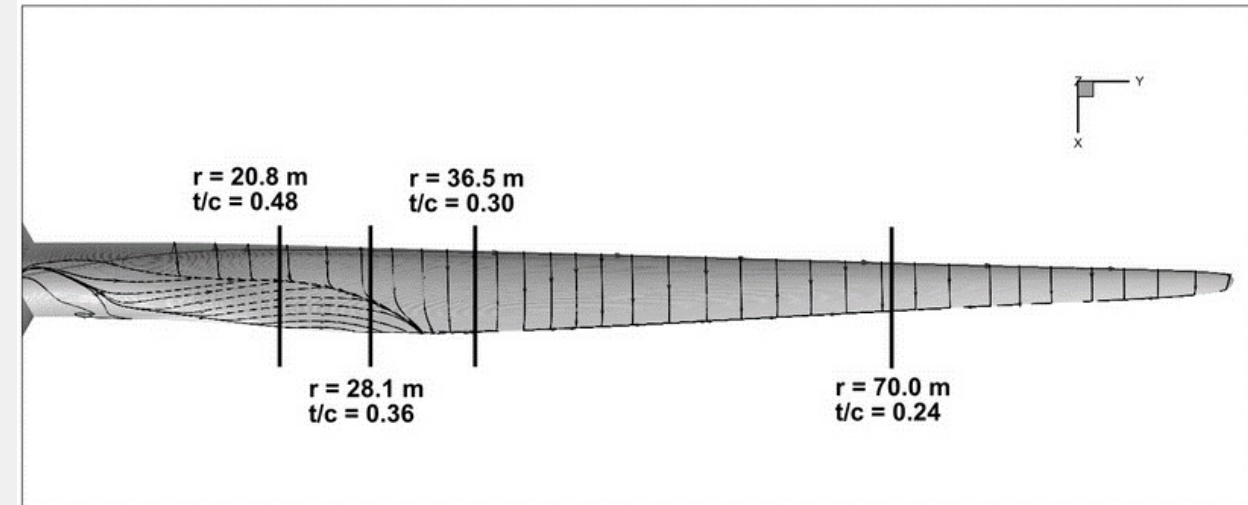
*Variation of Reynolds number on the DTU 10MW wind turbine blade operating at a wind speed of 10 m/s*

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Flow Over the Blade

For a commercial wind turbine,

- Compressibility
- Angle of attack
- Reynolds number
- Separation: Due to high angle of attack and very thick airfoils, separation may be present near the root.
- Laminar-turbulent transition:
  - For small wind turbines, laminar flow may be present
  - For large commercial wind turbines, (usually) fully-turbulent flow



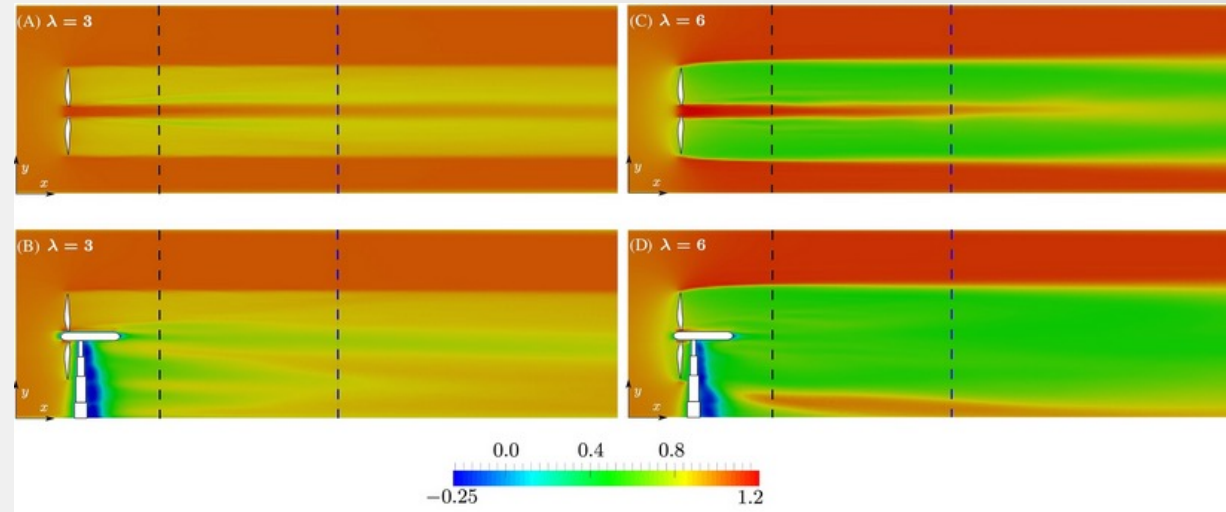
*Streamlines on the suction surface of DTU 10MW wind turbine blade operating at a wind speed of 10 m/s<sup>1</sup>*

<sup>1</sup>Frederik Zahle et al. "Comprehensive Aerodynamic Analysis of a 10 MW Wind Turbine Rotor Using 3D CFD". In: *32nd ASME Wind Energy Symposium*. Reston, Virginia: American Institute of Aeronautics and Astronautics, Jan. 2014. doi: 10.2514/6.2014-0359.

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Interactions with Turbine Components

- Spinner, nacelle, tower
- Differences in CFD applications:
  - Wake profile: Symmetrical/Asymmetrical
  - Jet in the center
  - Similar velocity profile over the blade<sup>2</sup>
    - Similar power and thrust for an **individual** turbine



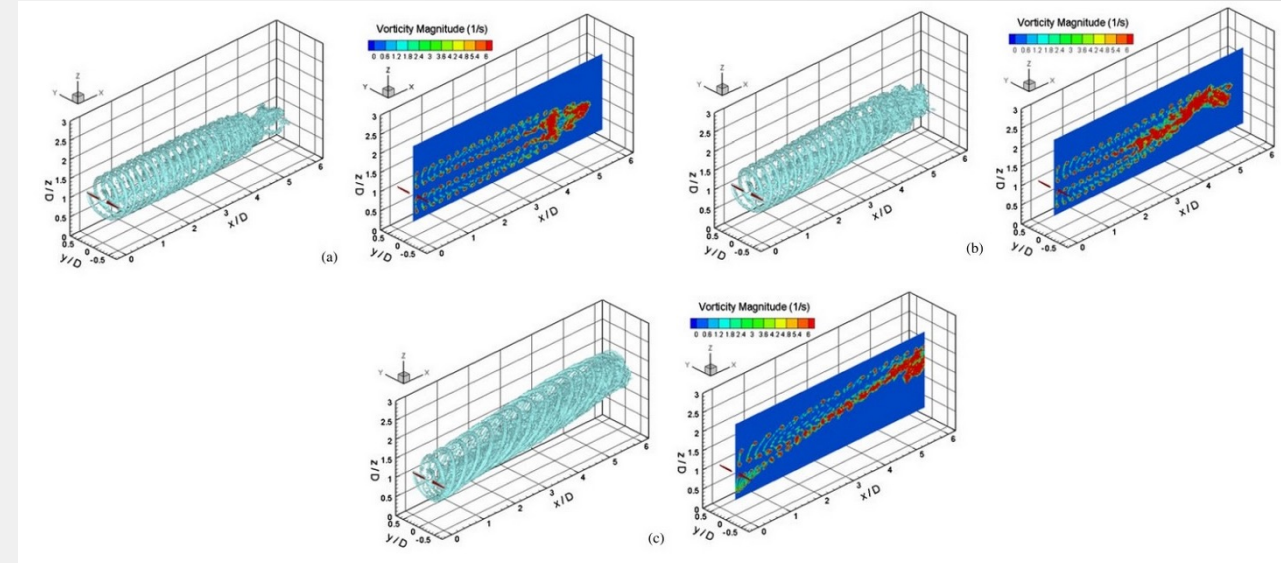
*Contours of streamwise velocity in a vertical section with different tip speed ratios and with (B,D) and without (A,C) tower and nacelle<sup>2</sup>*

<sup>2</sup>Christian Santoni et al. "Effect of tower and nacelle on the flow past a wind turbine". In: *Wind Energy* 20.12 (Dec. 2017), pp. 1927–1939. doi: 10.1002/we.2130

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Interactions with Non-turbine Elements

- Ground: complex terrain
  - Wind farm placement
- Sea: coupling with a hydrodynamic solver
- Atmospheric boundary layer (ABL):
  - Non-uniform inlet velocity profile
  - Affects the incoming flow velocity and turbulence
  - Impacts the wake and turbine performance
  - May not affect the turbine performance if the rotor is outside of ABL (some commercial WTs)



*Isovorticity surfaces and vorticity magnitude distributions within the wake after 10 revolutions on the  $y=0$  plane for different wind shear (ABL) cases: (a) uniform inflow, (b) normal wind profile, and (c) extreme wind shear<sup>3</sup>*

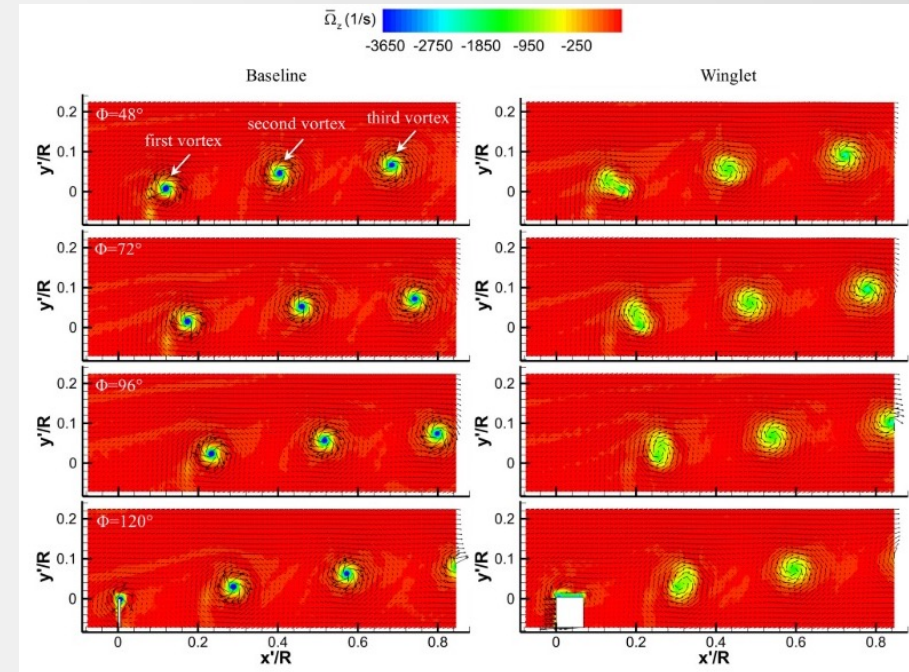
<sup>3</sup>Nilay Sezer-Uzol and Oguz Uzol. "Effect of steady and transient wind shear on the wake structure and performance of a horizontal axis wind turbine rotor". In: *Wind Energy* 16.1 (Jan. 2013), pp. 1–17. doi: 10.1002/we.514



# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Wakes

- Reduced velocity and increased turbulence
  - Affects the wind turbine performance
- Tip vortices
- Affected by the wind turbine control
- Unsymmetrical wakes
- Partial alignment



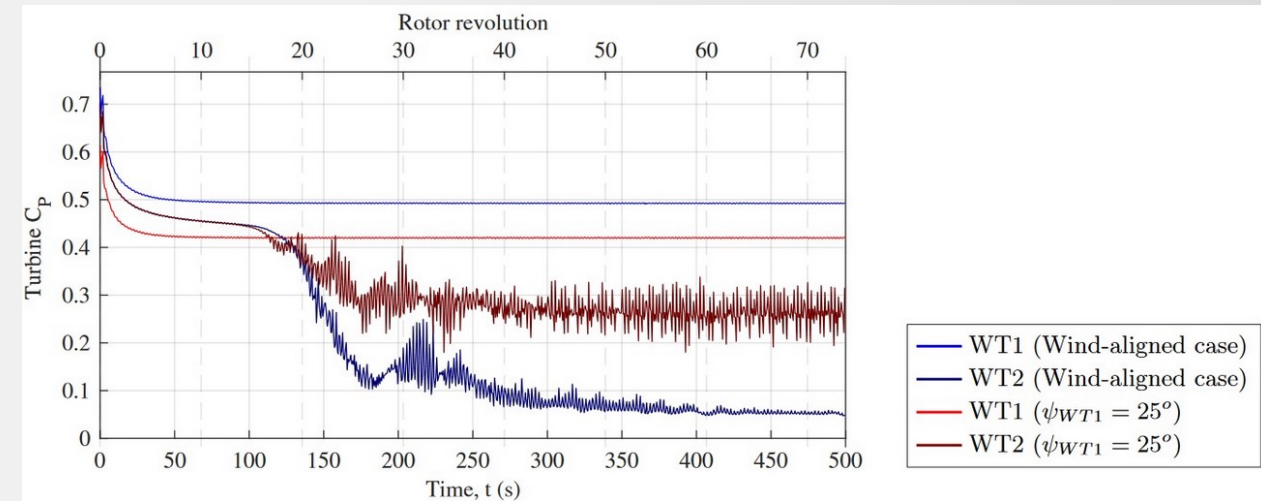
*Phase-averaged out-of-plane vorticity distributions for different rotor phases for baseline (left) and winglet (right) cases<sup>4</sup>*

<sup>4</sup>Yaşar Ostovan, M. Tuğrul Akpolat, and Oğuz Uzol. “Experimental investigation of the effects of winglets on the tip vortex behavior of a model horizontal axis wind turbine using particle image velocimetry”. In: *Journal of Solar Energy Engineering, Transactions of the ASME* 141.1 (2019), pp. 1–13. issn: 15288986. doi: 10.1115/1.4041154.

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Wakes

- Reduced velocity and increased turbulence
- Tip vortices
- Affected by the wind turbine control
- Unsymmetrical wakes
- Partial alignment



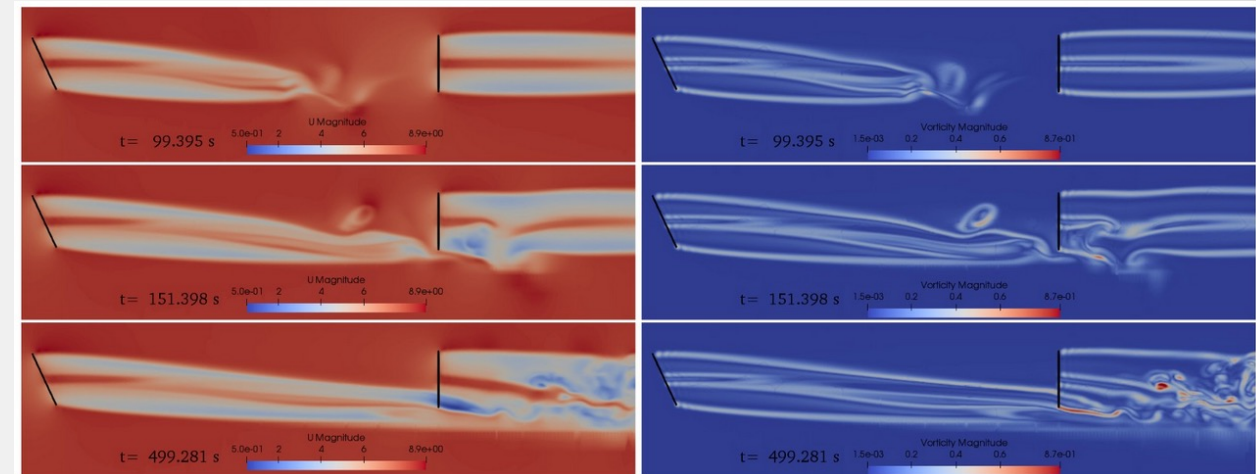
*Power coefficient histogram of upstream and downstream wind turbines with and without wind turbine control<sup>5</sup>*

<sup>5</sup>Huseyin C. Onel and Ismail H. Tuncer. "A comparative study of wake interactions between wind-aligned and yawed wind turbines using LES and actuator line models". In: *Journal of Physics: Conference Series* 1618.6 (Sept. 2020), p. 062009. doi: 10.1088/1742-6596/1618/6/062009

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Wakes

- Reduced velocity and increased turbulence
- Tip vortices
- Affected by the wind turbine control
- Unsymmetrical wakes
- Partial alignment



(a) Velocity contours

(b) Vorticity contours

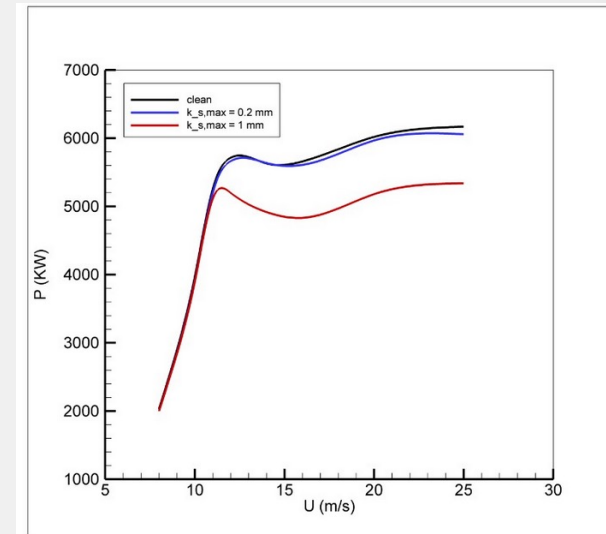
*Wake development behind a yawed upstream turbine<sup>5</sup>*

<sup>5</sup>Huseyin C. Onel and Ismail H. Tuncer. "A comparative study of wake interactions between wind-aligned and yawed wind turbines using LES and actuator line models". In: *Journal of Physics: Conference Series* 1618.6 (Sept. 2020), p. 062009. doi: 10.1088/1742-6596/1618/6/062009

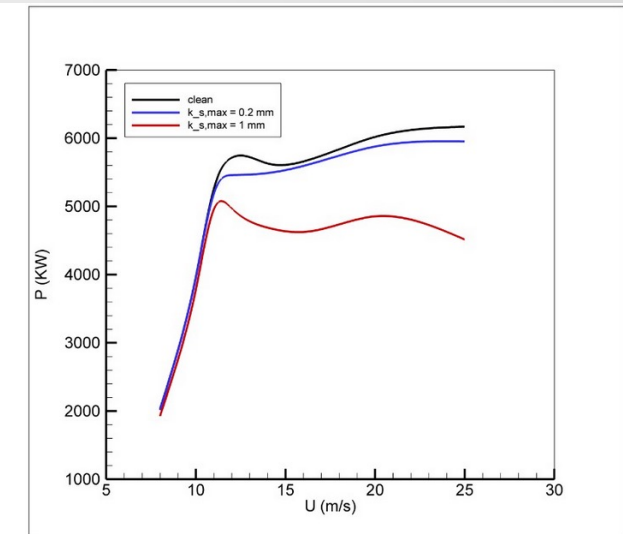
# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Additional Considerations

- Surface roughness:
  - Leading edge contamination and erosion
  - Deterioration of turbine performance
  - CFD applications:
    - Direct modelling through the mesh
    - Roughness models
- Aeroelasticity (Fluid-structure interaction)
- Wind farm control
- Gusts
- Aeroacoustics



(a) 40 microns



(b) 150 microns

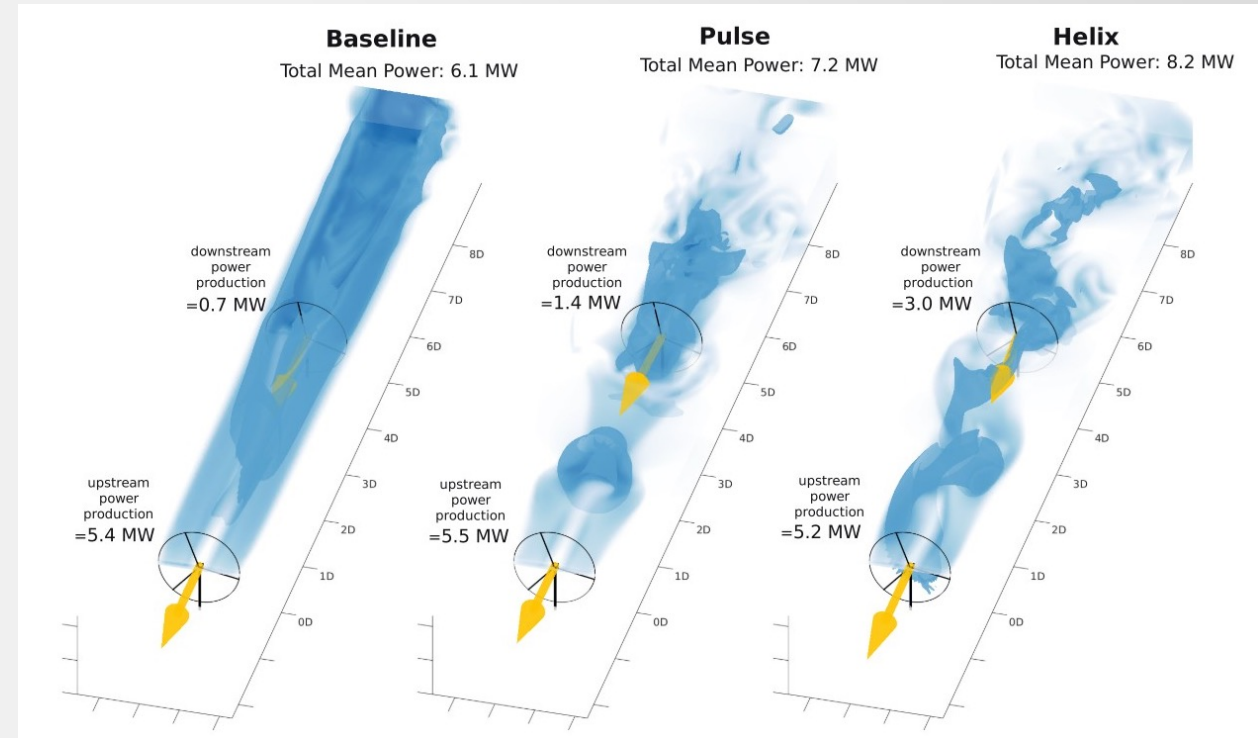
*Power output with contaminated turbine blades with different contaminant particle sizes<sup>6</sup>*

<sup>6</sup>Serkan Özgen, Eda Bahar Sarıbel, and Ali Rıza Yaman. "Effect of blade contamination on power production of wind turbines". In: *Journal of Physics: Conference Series* 2265.3 (May 2022), p. 032012. doi: 10.1088/1742-6596/2265/3/032012

# Wind Turbine Rotor Aerodynamics and CFD Implementations

## Additional Considerations

- Surface roughness
- Aeroelasticity
- Wind farm control:
  - Novel methods require high-fidelity CFD simulations
- Gusts
- Aeroacoustics

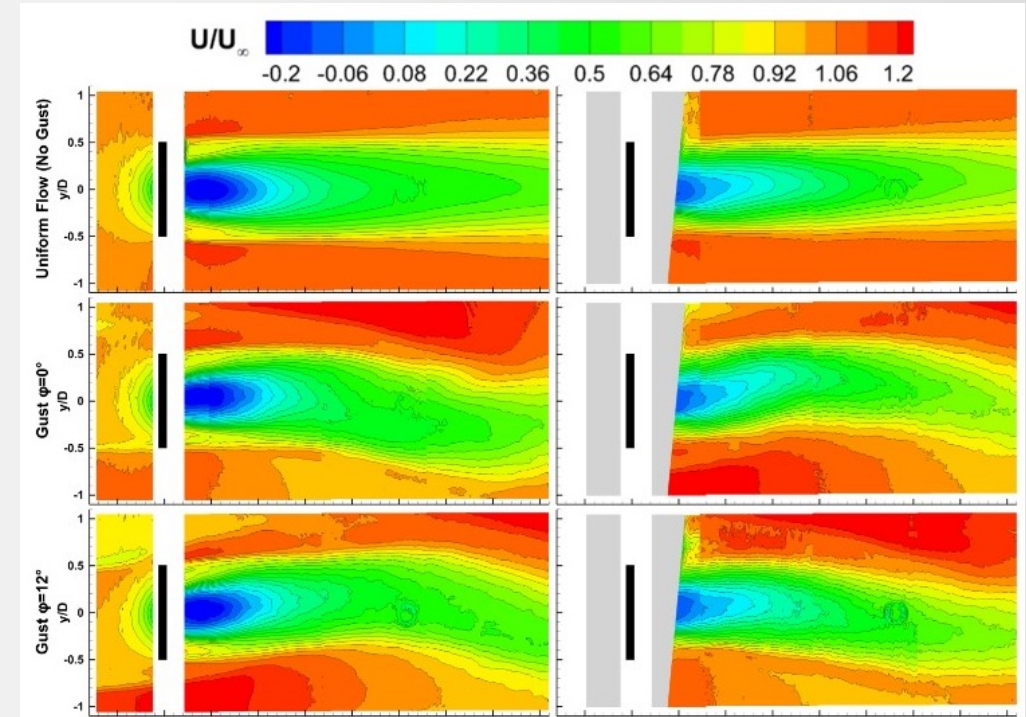


*Illustration from a LES study of a stable normal operation wake (left), wake resulting from (periodic) dynamic induction control (middle), wake resulting from the helix approach (right) along downstream distances<sup>7</sup>*

<sup>7</sup>Johan Meyers et al. “Wind farm flow control: prospects and challenges”. In: *Wind Energ. Sci. Discuss. [preprint]* (2022). doi: 10.5194/wes-2022-24.

# Wind Turbine Rotor Aerodynamics and CFD Implementations Additional Considerations

- Surface roughness
- Aeroelasticity
- Wind farm control
- Gusts:
  - brief increase in the speed of the wind
  - may reduce the effectiveness of wind turbines
- Aeroacoustics



*PIV measurement results of a porous disc (left) and model wind turbine (right) under uniform flow, porous disc under gusty inflow, when the gust vanes are at 0, 12 degrees, respectively<sup>8</sup>*

<sup>8</sup>İmge Yiğili et al. "Design of a gust generator and comparison of model wind turbine and porous disc wake flows in a transverse gust". In: *Journal of Physics: Conference Series* 2265.2 (May 2022), p. 022108. doi: 10.1088/1742-6596/2265/2/022108

## Content

1. Wind Turbine Rotor Aerodynamics and CFD Implementations
  1. Flow Over the Blade
  2. Interactions with Turbine Components
  3. Interactions with Non-Turbine Elements
  4. Wakes
  5. Additional Considerations
2. SU2 Capabilities and Tutorial
  1. Fundamentals
  2. Dynamic Mesh
  3. Flow Over the Blade
  4. Interactions with Rotor Components
  5. Actuator Disk
  6. Surface Roughness and Gusts
3. Applications
  1. Test Case Configuration
  2. SU2 Configuration
  3. Results
  4. Possible Improvements

# SU2 Capabilities and Tutorial Fundamentals



- Inputs:
  - Configuration file (through the SU2\_CFD command)
    - SU2\_CFD config\_template.cfg
    - `mpirun -n 16 SU2_CFD config_template.cfg`
    - Python scripts: `python parallel_computation.py -f config_template.cfg -n 16`
  - Mesh file (through the configuration file)
    - MESH\_FILENAME
    - MESH\_FORMAT = SU2 or CGNS

```
NDIME= 2
NELEM= 208896
```

9	0	1	386	385	0
9	1	2	387	386	1
9	2	3	388	387	2
9	3	4	389	388	3
9	4	5	390	389	4
9	5	6	391	390	5
9	6	7	392	391	6
9	7	8	393	392	7
9	8	9	394	393	8
9	9	10	395	394	9
9	10	11	396	395	10

*SU2 format mesh file*

```
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%
% SU2 configuration file
% Case description: _____
% Author: _____
% Institution: _____
% Date: _____
% File Version 7.4.0 "Blackbird"
%
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

% ----- DIRECT, ADJOINT, AND LINEARIZED PROBLEM DEFINITION -----%
%
% Solver type (EULER, NAVIER_STOKES, RANS,
%             INC_EULER, INC_NAVIER_STOKES, INC_RANS,
%             NEMO_EULER, NEMO_NAVIER_STOKES,
%             FEM_EULER, FEM_NAVIER_STOKES, FEM_RANS, FEM_LES,
%             HEAT_EQUATION_FVM, ELASTICITY)
%
% SOLVER= EULER
%
% Specify turbulence model (NONE, SA, SST)
%
% KIND_TURB_MODEL= NONE
```

*Configuration text file*

```
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%
% ----- INPUT/OUTPUT FILE INFORMATION -----%
%
% Mesh input file
% MESH_FILENAME= mesh_NACA0012_inv.su2
%
% Mesh input file format (SU2, CGNS)
% MESH_FORMAT= SU2
```

*Mesh file information within the SU2 configuration file*



# SU2 Capabilities and Tutorial Fundamentals

- SU2 uses an outer time loop to march through the physical time, and an inner loop which is usually a pseudo-time iteration or a (quasi-)Newton scheme.
- Time dependency is set by:
  - `TIME_DOMAIN = NO` : Steady simulation (Only an inner time loop)
  - `TIME_DOMAIN = YES` : Unsteady simulation
- The number of iterations are controlled by
  - Steady simulation: `ITER = 1000`
  - Unsteady simulation:
    - `TIME_STEP = 1.0`
    - `MAX_TIME = 50.0`
    - `INNER_ITER = 100`
    - `TIME_ITER = 200`
- Time marching methods for solving the differential equations:
  - First-order dual-time stepping: `TIME_MARCHING = DUAL_TIME_STEPPING-1ST_ORDER`
  - Second-order dual-time stepping: `TIME_MARCHING = DUAL_TIME_STEPPING-2ND_ORDER`
  - Conventional time stepping: `TIME_MARCHING = TIME_STEPPING`

Simulation time =  $\text{MIN}(\text{TIME\_ITER} * \text{TIME\_STEP}, \text{MAX\_TIME})$

# SU2 Capabilities and Tutorial Fundamentals

## Boundary conditions (BC):

- Constant heatflux wall, in other words no-slip condition:

- `MARKER_HEATFLUX = (Wall11, 1e05, Wall12, 0.0)`

Name of the BC    Heatflux in [W/m<sup>2</sup>]

Adiabatic wall

- Other no-slip conditions: `MARKER_HEATTRANSFER`, `MARKER_ISOTHERMAL`
- Farfield boundary condition (freestream values):

- `MARKER_FAR = (farfield)`

- Outlet boundary condition:

- `MARKER_OUTLET = (outlet, 0.0)`

- Compressible: ( outlet marker, back pressure (static thermodynamic), ... )
- Inc. Pressure: ( outlet marker, back pressure (static gauge in Pa), ... )
- Inc. Mass Flow: ( outlet marker, mass flow target (kg/s), ... )

## Boundary conditions (BC):

### • Inlet boundary condition:

- `INLET_TYPE = TOTAL_CONDITIONS` or `MASS_FLOW`
- `MARKER_INLET =`
  - **Total Conditions:** (inlet marker, total temp, total pressure, flow\_direction\_x, flow\_direction\_y, flow\_direction\_z, ...)
  - **Mass Flow:** (inlet marker, density, velocity magnitude, flow\_direction\_x, flow\_direction\_y, flow\_direction\_z, ... )
  - Inc. Velocity, Inc. Pressure
- One can give a non-uniform inlet profile through:
  - `SPECIFIED_INLET_PROFILE= YES`
  - `INLET_FILENAME= inlet.dat`
    - If SU2 cannot find a file with the `INLET_FILENAME` then it creates a template file for the mesh specified in the configuration file

## Boundary conditions (BC):

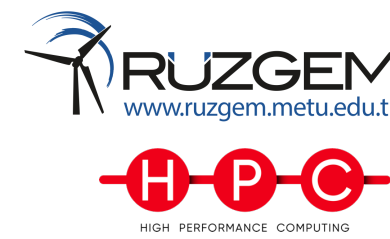
- Periodic boundary condition (e.g. 120° single-blade simulations):
  - `MARKER_PERIODIC= ( periodic marker, donor marker, rotation_center_x, rotation_center_y, rotation_center_z, rotation_angle_x-axis, rotation_angle_y-axis, rotation_angle_z-axis, translation_x, translation_y, translation_z, ...)`

## Output:

- To list the all possible output fields for the given configuration file, run the command:
  - `SU2_CFD -d config_template.cfg` (performs a dry run)
- Rotor application:
  - `ROTATING_FRAME` (figure of merit, thrust, torque) group within the screen/history output fields
  - `VORTEX_IDENTIFICATION` group within the volume output fields

# SU2 Capabilities and Tutorial

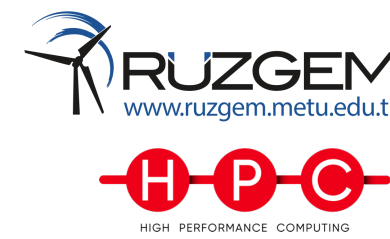
## Dynamic Mesh



- To define the type of dynamic mesh:
  - `GRID_MOVEMENT= RIGID_MOTION, ROTATING_FRAME, or STEADY_TRANSLATION`
  - `RIGID_MOTION`: Physically rotating the mesh
  - `ROTATING_FRAME`: Implementing the flow equations in the rotating coordinate frame with additional source terms
- Definitions:
  - `MACH_MOTION = 0.2` (Used for initializing and computing force coeffs. with dynamic meshes)
  - `MOTION_ORIGIN = 0.0 0.0 0.0`
  - `ROTATION_RATE = 0.0 0.0 5.0` (Angular velocity vector [rad/s])

# SU2 Capabilities and Tutorial

## Flow Over the Blade



- Transition models (valid for Version 7.4.0)
  - SA-BCM
    - `KIND_TURB_MODEL= SA`
    - `SA_OPTIONS= BCM`
    - `KIND_TRANS_MODEL= BC` (earlier than Version 7.3.1)
  - SST LM (Langtry-Menter Trans. SST)
    - `KIND_TURB_MODEL= SST`
    - `KIND_TRANS_MODEL= LM`
  - Turbulence intensity
    - `FREESTREAM_TURBULENCEINTENSITY= 0.05 (=5%)`
- Wall functions:
  - **Markers:** `MARKER_WALL_FUNCTIONS=(wall1, STANDARD_WALL_FUNCTION, wall2, EQUILIBRIUM_WALL_MODEL)`
  - The von Karman constant: `WALLMODEL_KAPPA`
  - The wall function model constant B: `WALLMODEL_B`
  - The  $y^+$  value below which the wall function is switched off: `WALLMODEL_MINYPPLUS`

# SU2 Capabilities and Tutorial

## Interactions with Rotor Components



- To include both rotating and non-rotating components, the user must specify different configuration files for those components.
- Main configuration file:
  - `MULTIZONE=YES`
  - `CONFIG_LIST = ( input_inner.cfg, input_outer.cfg )`
  - `MARKER_ZONE_INTERFACE= ( interface_inner, interface_outer )`
  - `MARKER_FLUID_INTERFACE= ( interface_inner, interface_outer )`
    - Each zone is discretized independently
  - `KIND_INTERPOLATION= NEAREST_NEIGHBOR, ISOPARAMETRIC, or SLIDING_MESH`
    - Specifies how values are communicated over the interfaces

# SU2 Capabilities and Tutorial

## Interactions with Rotor Components



- `input_inner.cfg`:
  - `GRID_MOVEMENT= RIGID_MOTION`
  - `MESH_FILENAME= rotor.su2`
- `input_outer.cfg`:
  - `GRID_MOVEMENT= NONE`
  - `MESH_FILENAME= outer_cylinder.su2`



# SU2 Capabilities and Tutorial

## Actuator Disk

- The actuator disk model in SU2 is not much suitable with wind turbines, but still usable

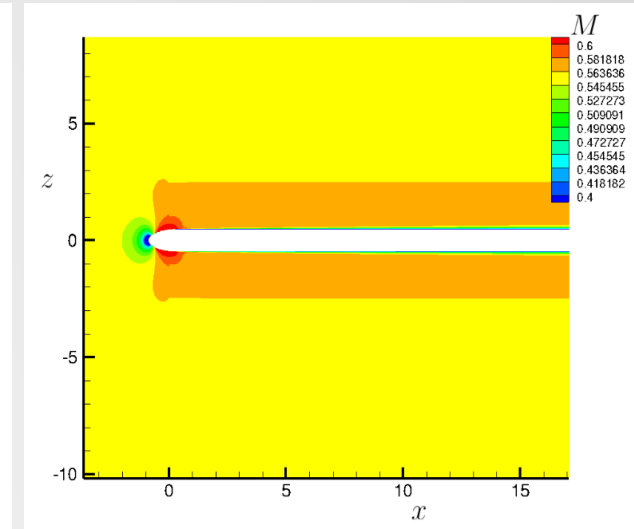
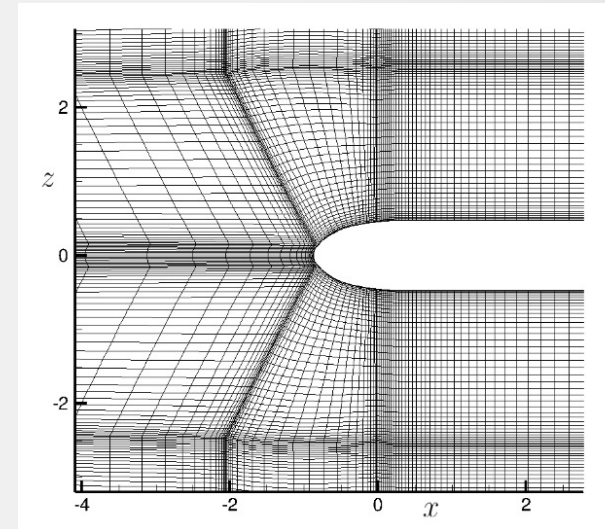
- **Commands:**

- `MARKER_ACTDISK= (inlet face marker, outlet face marker, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0)`
- `ACTDISK_TYPE= VARIABLE_LOAD`
- `ACTDISK_FILENAME= actuatordisk.dat`

```

MARKER_ACTDISK= DISK DISK_BACK
CENTER= 0.0 0.0 0.0
AXIS= 1.0 0.0 0.0
RADIUS= 2.5146
ADV_RATIO= 2.81487
NROW= 37
# rs=r/R    dCT/drs    dCP/drs    dCR/drs
0.2031    0.020066    0.0890674    0.0
0.2235    0.019963    0.0932674    0.0
0.2439    0.021707    0.0982980    0.0
0.2644    0.024667    0.1064153    0.0
  
```

Sample  
actuator  
disk file



Mesh details for the actuator disk with a semi-infinite spinner test case (left) and SU2 Mach number contour<sup>9</sup>

<sup>9</sup>E. Saetta, L. Russo, and R. Tognaccini. “Implementation and validation of a new actuator disk model in SU2”. In: *SU2 Conference 2020* (June 2020).

# SU2 Capabilities and Tutorial

## Surface Roughness and Gusts

- SU2 contains a model to simulate surface roughness.
- `WALL_ROUGHNESS = (wall1, ks1)`
  - `ks1` is the equivalent sand grain roughness height on each of the wall in meters.
- SU2 contains a gust model for top-hat, sine, 1-cos, vortex, and extreme operating gust (EOG) gust types.
  - **Only in 2D!** However, the developers are working on for a 3D model.
  - Commands:
    - `WIND_GUST = YES`
    - `GUST_TYPE = TOP_HAT, SINE, ONE_M_COSINE, VORTEX, or EOG`
    - `GUST_DIR = X_DIR or Y_DIR`
    - `GUST_WAVELENGTH= 10.0 (in meters)`
    - `GUST_PERIODS= 1.0 (Number of gust periods)`
    - `GUST_AMPL= 10.0 (in m/s)`
    - `GUST_BEGIN_TIME= 0.0`
    - `GUST_BEGIN_LOC= 0.0`

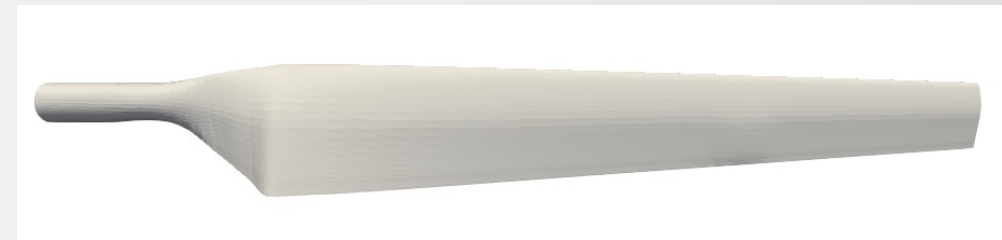
## Content

1. Wind Turbine Rotor Aerodynamics and CFD Implementations
  1. Flow Over the Blade
  2. Interactions with Turbine Components
  3. Interactions with Non-Turbine Elements
  4. Wakes
  5. Additional Considerations
2. SU2 Capabilities and Tutorial
  1. Fundamentals
  2. Dynamic Mesh
  3. Flow Over the Blade
  4. Interactions with Rotor Components
  5. Actuator Disk
  6. Surface Roughness and Gusts
3. Applications
  1. Test Case Configuration
  2. SU2 Configuration
  3. Results
  4. Possible Improvements

# Applications

## Test Case Configuration

- NREL Phase VI wind turbine
  - 2 blades
  - Rotor radius: 5.029 m
  - Rotational speed: 72 RPM = 7.54 rad/s
  - S809 airfoil starting at  $r/R=25\%$
- $u_{\infty} = 9$  m/s (uniform inlet)
- $\lambda = 4.2$  (tip speed ratio)

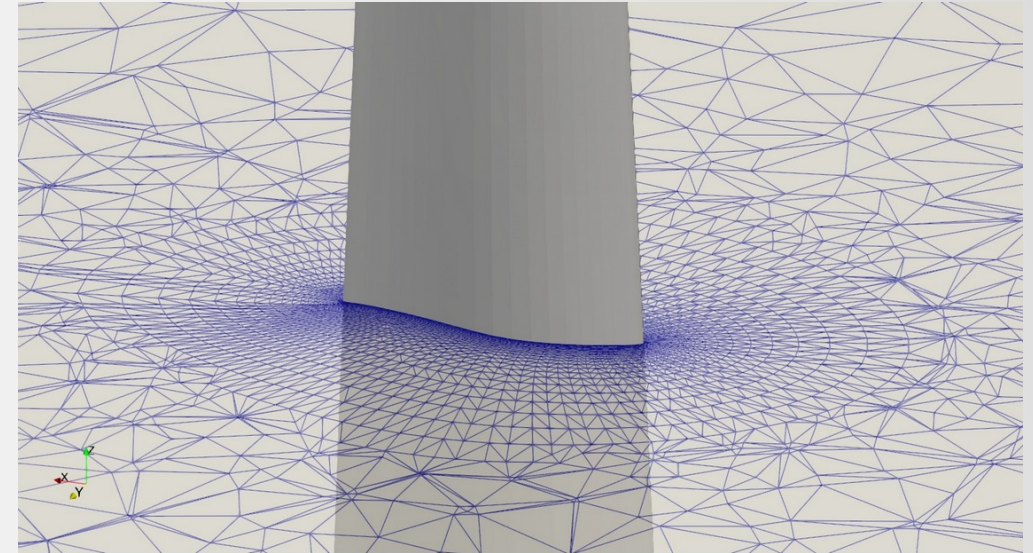


*NREL Phase VI wind turbine blade*

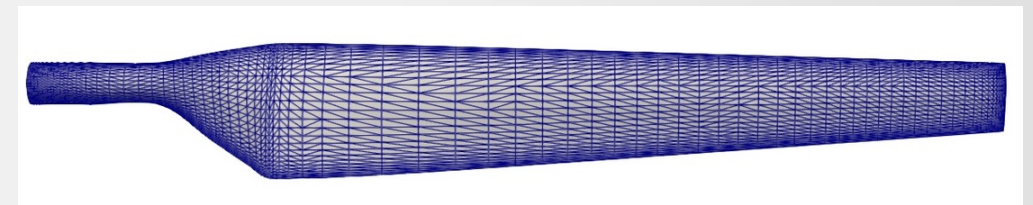
# Applications

## Test Case Configuration

- Single blade with periodic boundaries
- 800k cells and 10k surface cells
- Boundary layer with a first cell height of  $\Delta y = 1 \times 10^{-4}$  m and  $y_{max}^+ \approx 25$
- Boundaries:
  - Inlet (-x): 5D
  - Side (circular): 5D
  - Outlet (+x): 50D



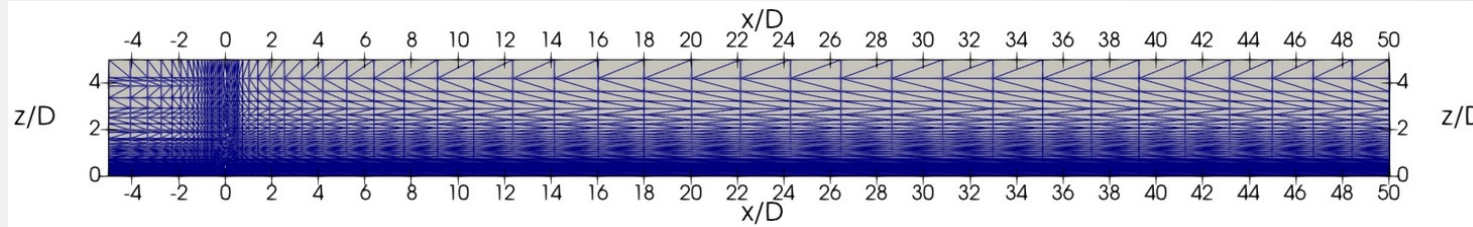
*Mesh on the z=3 m cross-section (check the mesh near the surface)*



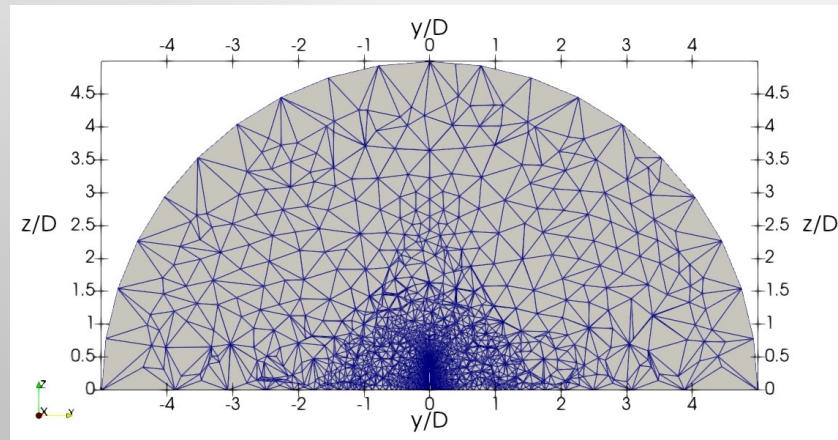
*Surface mesh*

# Applications

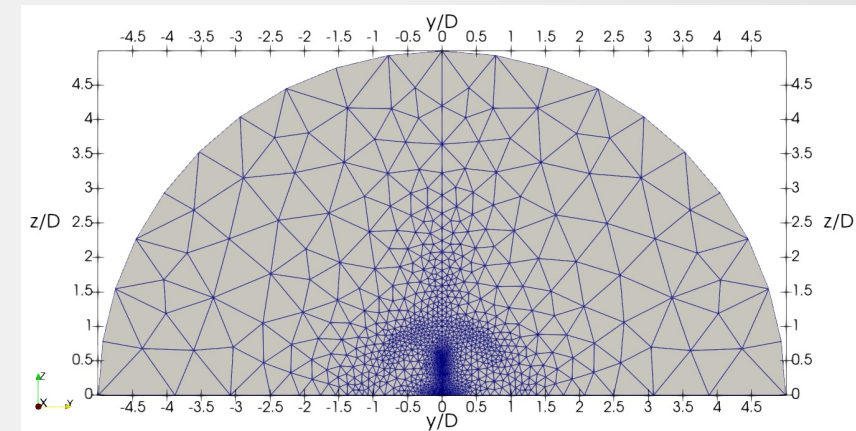
## Test Case Configuration



*Mesh on the  $y=0$  cross-section*



*Mesh on the  $x=0$  cross-section*



*Mesh on the  $x=0$  cross-section*

# Applications

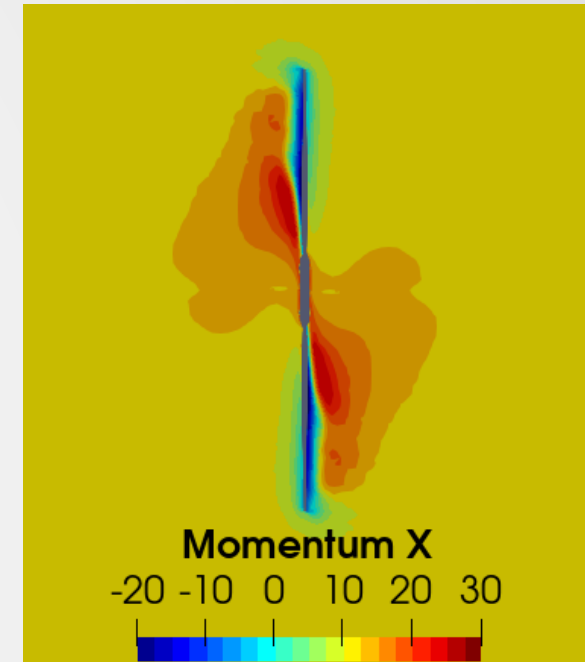
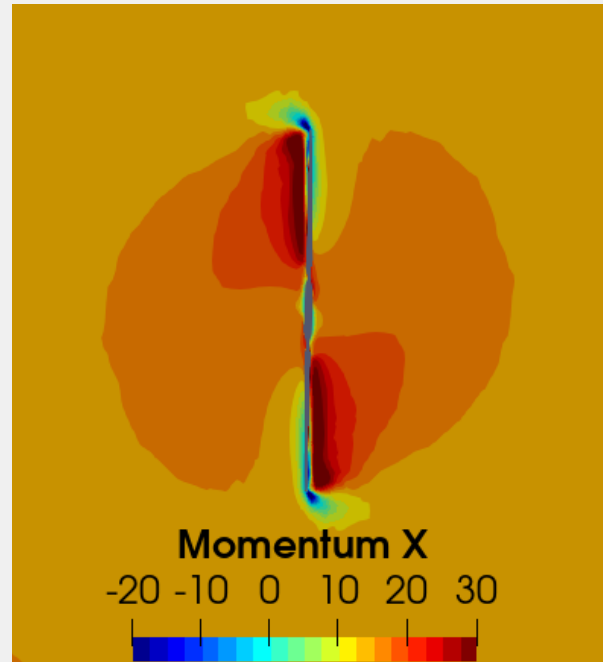
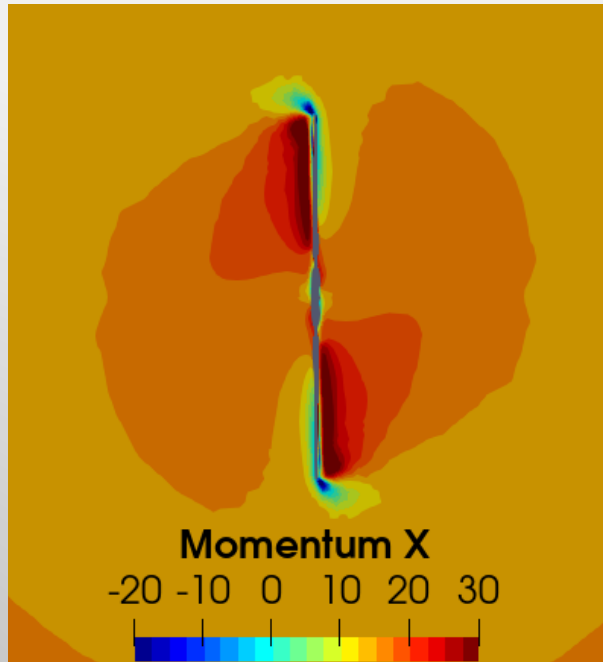
## SU2 Configuration



- 3 simulations:
  - Half (Single blade) configuration, rigid mesh motion, unsteady simulation
    - 10h on 16 cores for a 4 seconds (approx. 5 revolutions) simulations
  - Full (Two-blade) configuration, rigid mesh motion, unsteady simulation
    - 20h on 16 cores for a 4 seconds (approx. 5 revolutions) simulations
  - Half (Single blade) configuration, rotating frame, steady simulation
    - 2h on 16 cores for 12,000 iteration ( $\log(\text{RMS density}) = -3$  for this case)
- Shall we check the configuration files?

- Shall we check the output files?
  - history.csv
    - Variation of residuals, force coefficients, moment coefficients with time (iterations)
  - flow.vtu
    - Volume flow results
  - surface\_flow.vtu
    - Surface flow results
  - forces\_brakedown.dat
    - Force and moment brakedown by surface



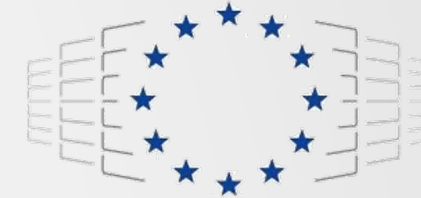


*Streamwise velocity variations along the blade from the rigid motion single blade (left), rigid motion two-blade (center), and reference frame (right) simulations*

- Shorter first cell height, so lower  $y^+$ 
  - If not, implementing the wall functions
- Transition model to capture the laminar flow and laminar-turbulent transition
- Remaining rotor components and non-rotor components for interactions
- Atmospheric boundary layer (non-uniform inlet velocity profile) since this is a small turbine (rotor within the ABL)
- Wind tunnel walls if there will be a validation against experimental data
- Finer mesh around the blade to preserve the turbulent structures

METU RÜZGEM  
HPC@ODTÜ  
EuroCC@Türkiye

<https://ruzgem.metu.edu.tr>  
<https://www.hpc.info.tr>  
<https://eurocc.truba.gov.tr>



**EuroHPC**  
Joint Undertaking

This project has received funding from the European High-Performance Computing Joint Undertaking (JU) under grant agreement No 951732. The JU receives support from the European Union's Horizon 2020 research and innovation programme and Germany, Bulgaria, Austria, Croatia, Cyprus, Czech Republic, Denmark, Estonia, Finland, Greece, Hungary, Ireland, Italy, Lithuania, Latvia, Poland, Portugal, Romania, Slovenia, Spain, Sweden, United Kingdom, France, Netherlands, Belgium, Luxembourg, Slovakia, Norway, Switzerland, Turkey, Republic of North Macedonia, Iceland, Montenegro